GREAT DESIGNS IN

EVALUATION OF DAMAGE ACCUMULATION FRACTURE MODELS IN NON-LINEAR STRAIN PATHS

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PROJECT TEAM

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OVERVIEW OF FRACTURE MODELLING FOR CRASH

1. Construct Proportional Fracture Loci from Coupon Tests: Experimental or Inverse FE Approaches Used



0 8

3-Point Bend of Rail Section

Step 2. Assume Damage Model for Non-Linear Loading. Pair with Fracture Locus

$$\Delta D^{GISSMO} = \left[\frac{n}{\varepsilon_{f}^{\exp}\left(T\right)} D^{\left(1-\frac{1}{n}\right)}\right] \Delta \varepsilon^{p}$$

Step 3. Regularize for element size & apply to structural CAE models

Need Objective Evaluation of Phenomenological Fracture Models



Axial Crush of Front End Crush Structure



Stress Triaxiality



Axial Crush of Hot Stamped Rail

BRIEF REVIEW OF PROPORTIONAL FRACTURE

Example: Consider isotropically hardening material. Yield surface expands Fracture: Define failure surface: Tresca is simplest with straight line for fracture Intersection of yield and fracture surface creates the fracture locus (2D) or surface (3D)



 $(\sigma_1)^f$ = constant (plane stress, tensile quadrant)

BRIEF REVIEW OF PROPORTIONAL FRACTURE

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Change Tresca to Mohr-Coloumb for Difference in Tension and Compression (still linear fracture model)

Mohr-Coulomb Criterion:

 $\sigma_1 - \sigma_3 + c(\sigma_1 + \sigma_3) = b$

b = related to shear stress;

c = cohesion (controls compressive response) c = 0 (Tresca – Max shear stress criterion)

C parameter acts as hinge at uniaxial tension. Cusp in plane stress fracture locus created by corner

Problem Statement: Identify the Fracture Potential Function in Stress Space. Natural extension to anisotropy, complex loading



BRIEF REVIEW OF PROPORTIONAL FRACTURE

Inversion of fracture surface with yield and hardening models used to create equivalent strain representation with triaxiality and Lode parameter. Ex: MMC-5 parameter model

$$\varepsilon_{f}^{MMC5} = \left[c_{2}\left[c_{\theta}^{s} + \frac{\sqrt{3}}{2 - \sqrt{3}}\left(c_{\theta}^{ax} - c_{\theta}^{s}\right)\left(\sec\left(\frac{\pi\overline{\theta}}{6}\right) - 1\right)\right]\left(\frac{\sqrt{3}}{3}\cos\left(\frac{\pi\overline{\theta}}{6}\right) + c_{1}\left[T + \frac{1}{3}\sin\left(\frac{\pi\overline{\theta}}{6}\right)\right]\right)\right]^{\frac{-1}{c_{n}}} \qquad T = \frac{\sigma_{hyd}}{\sigma_{eq}^{von\,Mises}} \qquad \overline{\theta} = 1 - \frac{2}{\pi}\cos^{-1}\left(\frac{27}{2}\frac{J_{3}}{\left(\sigma_{eq}^{vM}\right)^{3}}\right)$$

Convenient but breaks explicit link between plasticity and fracture surface. Now many versions in the literature

- Procedure for Experimental Fracture Characterization
- 1. Characterize plasticity and hardening behavior
- 2. Characterize fracture strains in proportional loading → what tests to use?
- 3. Select & calibrate fracture function
 → Shape can vary between calibration points...
- 4. How to generalize to non-proportional model
 → need damage model if using eq. strain (ex: GISSMO)



ANISOTROPIC PLASTICITY

Detailed experimental characterization of anisotropy:

- Uniaxial tension (7 directions)
- Simple shear (3 directions)
- Plane Strain tension (3 directions)
- Disc compression test for biaxial R-value

Non-associated Yld2018 (Drucker-type) yield & plastic potential

1.2 1 1 0.8 1 0.6 1	$\sigma_{an} = \frac{1}{2} \sum_{n=1}^{3} \left[(J_{2}^{\prime(m)})^{3} - c (J_{2}^{\prime(m)})^{2} \right]^{1/6}$	1.16 1.14 1.12	Predicted Biaxial Stress Ratio: 1.014 Plane Strain Tension	1.1 -	Biaxial R-value Ratio: Model: 0.94 Exp. 0.94 ± 0.05	nent
0.4 - 0.2 - 0	c=1.226 BCC	1.1 1.08 1.06	Experiment	1		7
-0.4 -0.6 -0.8	von Mises	1.04	Uniaxial Tension	0.85		
-1 -1.2 -1.2	DP1180 Steel -1 -0.8 -0.4 -0.2 0 0.2 0.4 0.6 0.8 1 1.2 Normalized Stress: Rolling Direction	0.96	DP1180 Steel 10 20 30 40 50 60 70 80 90 Angle of Major Principal Stress with respect to Rolling Direction (°)	0.75 ¹ C	DP112 0 10 20 30 40 50 60 70 8 Angle of Major Principal Stress with respect to Rolling Direct	0 Steel

	Yld2018 Yield Function	c1 ⁽¹⁾	c2 ⁽¹⁾	c ₃ ⁽¹⁾	c ₆ ⁽¹⁾
		1.6080	1.2208	-1.2501	-0.1263
		c1 ⁽²⁾	c2 ⁽²⁾	c ₃ ⁽²⁾	c ₆ ⁽²⁾
		1.8253	2.2858	0.9587	2.2471
		c1 ⁽³⁾	c2 ⁽³⁾	c ₃ ⁽³⁾	c ₆ ⁽³⁾
		1.8204	1.8062	3.8849	2.5860
	Yld2018 Plastic Potential	c1 ⁽¹⁾	c2 ⁽¹⁾	c ₃ ⁽¹⁾	c ₆ ⁽¹⁾
		-1.4247	-1.1807	-2.5197	-1.7997
		c1 ⁽²⁾	c ₂ ⁽²⁾	c ₃ ⁽²⁾	c ₆ ⁽²⁾
		-2.3117	-3.4206	-2.1952	-2.7765
		c1 ⁽³⁾	c2 ⁽³⁾	c ₃ ⁽³⁾	c ₆ ⁽³⁾
		-1.4638	-0.7428	-0.4724	-0.7330

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CONSTITUTIVE CHARACTERIZATION

Shear hardening behavior nearly identical in RD-TD and TD-RD shear loading (orthotropic yield surface)

- \rightarrow Reverse shear test did not show complex hardening response. Reversion to monotonic shear response
- \rightarrow Characterization of hardening in complex strain path changes outside of scope \rightarrow future work



ISOTROPIC HARDENING BEHAVIOR

Hardening to large strains obtained using UW's shear conversion methodology in RD & TD

- → Uniform elongation is ~5% but shear conversion provides data to 60% plastic strain
- \rightarrow Isotropic hardening reasonable assumption for DP1180



Abedini A, Noder J, Kohar C, Butcher C, Mechanics of Materials, 2020, 148:103419

EVALUATIONS OF PLASTICITY MODEL



2.0

FORMABILITY IN TRANSVERSE DIRECTION

Forming limit curve (FLC) required to identify pre-straining limits for fracture tests Marciniak and Nakazima tests performed in Transverse direction (limiting direction)

- ightarrow Process corrections for Nakazima converged to Marciniak limit strains
- \rightarrow Modified Maximum Force Criterion (MMFC) predicted the FLC. No calibration parameters!



Nakazima Limit Strains

wardiniak & wakazima Limit Strains

Butcher et al. (2021). Journal of Materials Processing Technology. 287, 1-18

Min, J., Stoughton, T. B., Carsley, J. E., Lin, J. (2016). International Journal of Mechanical Sciences, 117, 115-134.

Hora P, Tong L, Reissner J, Modified maximum force criterion, a model for theoretical prediction of forming limit curves, Int J Mater Form, 2013, 6:267-279.

PROPORTIONAL FRACTURE: SIMPLE SHEAR

Shear fracture strains extremely sensitive to DIC settings and gage length Detailed study in Khameneh et al. (2022)

- \rightarrow Used DIC gage length of 0.50 mm throughout the study
- \rightarrow Fracture appeared to occur in center of gauge region based on void damage



Void formation in Mini-shear prior to Fracture



Void formation in Combined Tension & Shear Test: Average Triaxiality ~ 0.20



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PROPORTIONAL FRACTURE: PLANE STRAIN

V-bend test (VDA238-100) most reliable for plane strain fracture characterization

- \rightarrow V-bend provides proportional plane strain tension without necking
- \rightarrow DIC is inaccurate for Plane strain tests with necking. Stress state is triaxial...
- \rightarrow Necking-based tests can be improved with thickness strain correction





Fracture Strain at VDA load threshold





PROPORTIONAL FRACTURE: BIAXIAL STRETCHING

Biaxial fracture strains are significantly affected by necking

- \rightarrow Smaller punch radii suppress necking and create linear strain path to much higher strains
- ightarrow Apparent linearity can be misleading. Incremental analysis shows localization
- → Nakazima (R = 50 mm), Marciniak and Bulge tests tend to be lower bounds for biaxial strains in AHSS



PROPORTIONAL FRACTURE: UNIAXIAL TENSION

Uniaxial tension might be most challenging test due to necking instability and its type of localization Tensile geometry affects localization mode... the hole tension failed behind edge for DP1180 \rightarrow DIC fracture strains in necking-based samples require thickness correction or are too conservative Conical hole expansion (R = 5 mm) with machined hole gave best estimate for DP1180. No necking Hole expansion methodology and FE verification in Narayanan et al. (2022)



Specimen geometries of (a) JIS No.5. (b) miniature tensile, (c) tapered sub-sized ASTM-E8 tensile (d) central hole tension, and (e) schematic of hole expansion test with machined hole.



Major Principal Strain



CONICAL HOLE EXPANSION FOR UNIAXIAL FRACTURE

Conical hole expansion induces through-thickness strain gradient to suppress necking

Fracture initiates at outer edge that is in uniaxial tension for entire test

Use outer radius to fracture location with image processing. Easy!









Methodology to Characterize Fracture Strains

150

100

Hole expansion ratio based on outer diameter (%)

1.0

0.9

081140 0.7

0.6 blastic strain

Eduivalent 0.2

0.1

0.0

0

50



0.0

250

200

PROPORTIONAL FRACTURE LOCUS

Proportional fracture model calibrated using tests <u>without necking</u>: Approx. constant triaxiality in plane stress All tests analyzed with a consistent virtual strain gauge length (VSGL) of 0.5 mm.



FRACTURE IN BI-LINEAR STRAIN PATHS

How do the fracture strains vary in non-proportional loading?

In-Plane Pre-straining in uniaxial, plane strain and biaxial conditions

Select secondary fracture tests without necking and approx. linear strain paths

Major experimental effort: > 250 tests performed. Subset of data provided for NUMISHEET 2022 Benchmark



FIRST STAGE: IN-PLANE PRE-STRAINING

Uniaxial Pre-Straining: Oversized rectangular samples (strain variation < 1%) Plane Strain & Biaxial Pre-Straining: Marciniak tests (strain variation of 2% or less)

Local DIC strain history tracked for mapping to secondary fracture tests



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SELECT NLSP DATA: V-BEND AFTER PLANE STRAIN STRETCH

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Fracture strains constant for plane strain stretch + plane strain bend

Convergence – highlights plane strain fracture was accurately characterized



SELECT NLSP DATA: V-BEND AFTER BIAXIAL STRETCH

Repeatable strain paths and fracture strains

Comparable major principal strain at fracture in different strain paths

Cumulative equivalent fracture strain increases with biaxial pre-straining



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SELECT NLSP DATA: SHEAR AFTER BIAXIAL STRETCH

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Repeatable strain paths and fracture strains

Cumulative equivalent fracture strains in shear increase with biaxial stretch



DAMAGE ACCUMULATION AND FRACTURE MODELS: APPLYING PROPORTIONAL FRACTURE DATA TO NON-LINEAR STRAIN PATHS

All models are equally valid in proportional loading... but what happens in non-proportional loading? Stress-based representations do not require a damage counter: $\sigma_1 > \sigma_1^f(T,L)$

Equivalent strain representations employ phenomenological damage models (no physical foundation)



Linear damage (Johnson & Cook, 1983) commonly used

$$D = \int \frac{d\varepsilon_{eq}}{\varepsilon_f(T,L)} \qquad Fracture: D = 1$$

Power Law Damage (Xue, 2007)

$$\frac{\Delta D}{\Delta \varepsilon_{eq}^{p}} = \frac{n_{D}}{\varepsilon_{f}(T,L)} \left(\frac{\varepsilon_{eq}^{p}}{\varepsilon_{f}(T,L)} \right)^{n_{D}(1-1/n_{D})} D^{prop.} = \left(\frac{\varepsilon_{eq}^{p}}{\varepsilon_{f}(T,L)} \right)^{n_{D}}$$

GISSMO: Based on Power Law but implemented differently

$$\frac{\Delta D^{GISSMO}}{\Delta \varepsilon_{eq}^{p}} = \frac{n_{D}}{\varepsilon_{f}(T,L)} \frac{D^{(1-1/n_{D})}}{D_{initial} \neq 0,}$$
LS-DYNA uses $D_{initial} = 1.0e^{-20}$

Neukamm, F., Feucht, M., Haufe, A. Consistent damage modelling in the process chain of forming to crashworthiness simulations. LS-DYNA Anwenderforum, Bamberg 2008 Xue, L., (2007). International Journal of Solids and Structures 44,

DAMAGE ACCUMULATION EXAMPLE: DP1180

In-Plane Biaxial Pre-strain to Plane Strain Bending (change from highest to lowest fracture strains) GISSMO overestimated fracture. Linear slightly better than non-linear with recommended n = 2 Power Law damage was conservative. Influence of exponent can be different than GISSMO



DAMAGE ACCUMULATION EXAMPLE 2: DP1180

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Plane Strain Pre-Strain followed by Biaxial punch (path change from lowest to highest fracture strains) GISSMO underestimated fracture. Non-linear did better by reducing influence of first path in plane strain Power Law Damage similar to GISSMO when going from severe-to-mild stress state



EVALUATION OF FRACTURE LOCUS: GISSMO

Evaluate all non-linear strain path tests: 3 Fracture Loci with 4 parameters calibrated from 4 data points

Paired with GISSMO damage model using n = 2 Define correlation metric to evaluate model predictions





EVALUATION OF FRACTURE LOCUS: GISSMO*

0.2

0.3 0

Select Generalized Drucker-Prager model (marginally better)

MMC model rather conservative for second stage biaxial loading

Drucker-Yoon model overestimated fracture in second stage shear

$$D = \frac{\varepsilon_{eq-\text{Predicted}}^{f}}{\varepsilon_{eq-\text{Test}}^{f}} \qquad D = 1 \quad \text{Perfect correlation}$$



	GISSMO Damage Model: Damage Exponent = 2								
Dam	Modified Mohr-Coulomb Model		Gen. Drucker	Prager Model	Mod. Drucker-Yoon Model				
Fracture Metric: $\boldsymbol{\mathcal{E}}_{f}^{\text{model}} / \boldsymbol{\mathcal{E}}_{f}^{\text{exp}}$		Average	Std. Dev	Average	Std. Dev	Average	Std. Dev		
	Plane Strain Tension (12 Paths)	1.08	± 0.08	1.09	± 0.08	1.11	± 0.09		
Socondary Dath	Biaxial Stretching (9 Paths)	0.89	± 0.02	0.95	± 0.02	1.02	± 0.02		
Secondary Palli	Shear (12 Paths)	0.98	± 0.09	0.97	± 0.08	1.13	± 0.12		
	Uniaxial Tension (12 Paths)	0.93	± 0.07	0.93	± 0.06	0.94	± 0.06		

Equal-Biaxia

Tension

 $d\varepsilon_2$ $\rho = -$

 $d\varepsilon_1$

0.9

 $0.40 \le \rho \le 0.94$

 $0.65 \le T \le 0.66\overline{6}$

0.6

Principal Strain Ratio

DP1180-TD: Biaxial R-value

0.7

0.8

Asymptotic behavior in biaxial tension

- \rightarrow Triaxiality has poor resolution
- \rightarrow Consider additional biaxial tests and use NLSP due to necking as evaluation

*Results also depend upon choice of damage model afta calibration of loci

Triaxiality

FRACTURE REPRESENTATIONS: DP1180



Stress in Rolling Direction (MPa)

Stress Triaxiality at Fracture

GISSMO DAMAGE MODEL EVALUATION

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GISSMO Damage Accumulation:

$$D = \int \frac{n_D}{\varepsilon_f(T)} D^{(1-1/n_D)} d\varepsilon_{eq}^p$$

Assume Damage exponent : $n_D = 1, 2$

<u>Advantages</u>: Widely used and available in LS-DYNA, Linear form = Johnson-Cook model Eq. Strain based (relatively convenient)

<u>Disadvantages</u>: No physical foundation – user assumes damage exponent;

Damage accumulation unrelated to hardening ability

	Damage Model	GISSMO					
	Damage Metric & Parameters	Eq. S Damage Ex	itrain Iponent = 1	Eq. Strain Damage Exponent = 2			
Fract	ure Metric: $\mathcal{E}_{f}^{\text{model}} / \mathcal{E}_{f}^{\text{exp}}$	Average	Std. Dev	Average	Std. Dev		
	Plane Strain Tension (12 Paths)	1.05	± 0.08	1.09	± 0.08		
Sacandary Dath	Biaxial Stretching (9 Paths)	0.91	± 0.02	0.95	± 0.02		
Secondary Path	Shear (12 Paths)	0.94	± 0.08	0.97	± 0.08		
	Uniaxial Tension (12 Paths)	0.90	± 0.06	0.93	± 0.06		

<u>Observations:</u> Recommended non-linear damage (n = 2) performed better than linear damage (Johnson-Cook)

- \rightarrow Linear damage more conservative and will predict more fracture in FEA
- → Non-linear GISSMO over-predicted when switching to path with lower fracture strain (localization...)

POWER LAW DAMAGE MODEL EVALUATION

Power Law Damage Accumulation:

$$D = \int \frac{n_D}{\varepsilon_f(T)} \left(\frac{\varepsilon_{eq}^p}{\varepsilon_f(T)} \right)^{n_D - 1} d\varepsilon_{eq}^p$$

Assume Damage exponent : $n_D = 1, 1.5, 2$

Advantages: Appears to be consistent version of damage model GISSMO was based upon

Equivalent to GISSMO and Johnson-Cook for Linear Damage

Disadvantages: Same as GISSMO - No physical foundation – user assumes damage exponent

Damage accumulation unrelated to hardening ability

Damage Model		GISSMO				Power Law Damage					
Damage Metric & Parameters		Eq. Strain Damage Exponent = 1		Eq. Strain Damage Exponent = 2		Eq. Strain Damage Exponent = 1		Eq. Strain Damage Exponent = 1.5		Eq. Strain Damage Exponent = 2	
Fracture Metric: $\mathcal{E}_{f}^{\text{model}} / \mathcal{E}_{f}^{\text{exp}}$		Average	Std. Dev	Average	Std. Dev	Average	Std. Dev	Average	Std. Dev	Average	Std. Dev
	Plane Strain Tension (12 Paths)	1.05	± 0.08	1.09	± 0.08	1.05	± 0.08	0.99	± 0.05	0.96	± 0.05
Secondary Dath	Biaxial Stretching (9 Paths)	0.91	± 0.02	0.95	± 0.02	0.91	± 0.02	0.95	± 0.02	0.96	± 0.03
Secondary Path	Shear (12 Paths)	0.94	± 0.08	0.97	± 0.08	0.94	± 0.08	0.94	± 0.09	0.94	± 0.09
	Uniaxial Tension (12 Paths)	0.90	± 0.06	0.93	± 0.06	0.90	± 0.06	0.93	± 0.06	0.94	± 0.06

<u>Observations</u>: Performed superior to GISSMO with n = 1.5 - 2.0 recommended for DP1180

- → Significant improvement in predicting failure when second path is more severe as in localization
- → Overall, slightly conservative across all NLSP while GISSMO was not

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CONCLUSIONS AND FUTURE WORK

Fracture Characterization

- \rightarrow <u>Accurate plasticity model is required to differentiate stress states</u>
- \rightarrow Select proportional characterization tests without necking for fracture calibration
- → Choice of fracture loci and calibration is important: Biaxial region is critical.



Fracture Models: (Targeted for GDIS2024)

- \rightarrow Linear damage relatively conservative for fracture of DP1180
- → Power law damage model superior to GISSMO for non-linear damage
- ightarrow Investigate anisotropic fracture and application to higher ductility steel
- → Consider physically-based alternate models: Stress and/or plastic work based

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ADDITIONAL INFORMATION: PUBLICATIONS FOR DP1180 GDIS

[1] Abedini, A., Noder, J., Kohar, C.P., Butcher, C. (2020). Accounting for Shear Anisotropy and Material Frame Rotation on Constitutive Characterization of Automotive Alloys. *Mechanics of Materials*, 148, 1-17.

[2] Butcher, C., Jeyranpourkhameneh, J., Abedini, A., Connolly, D., Kurukuri, S. (2021). On the Experimental Characterization of Sheet Metal Formability and the Consistent Calibration of the MK Model for Biaxial Stretching in Plane Stress. *Journal of Materials Processing Technology*, 287, 1-18.

[3] Khameneh, F., Abedini, A, Butcher, C. (2021). Lengthscale effects in optical strain measurement for fracture characterization in simple shear. *International Journal of Fracture* 232, 153-180

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[8] Fast-Irvine, J.C. (2022). Experimental Methods for the Constitutive Characterization of Sheet Materials in Generalized Plane Strain. MASc Thesis. Available online to download.

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